

Undergraduate Certificate in Advanced Combustion Engineering (Part II) (United Kingdom)

Chemical Kinetics

Chemical kinetics is the branch of physical chemistry that studies the rates of chemical reactions and the factors that influence them. In advanced combustion engineering, a solid grasp of kinetic terminology is essential because the performance, stability, and emissions of combustion systems are directly linked to the underlying reaction pathways and their speed. The following exposition defines the principal terms and concepts encountered in the study of combustion kinetics, illustrating each with examples drawn from typical hydrocarbon flame environments, and highlighting the practical challenges that engineers face when applying these ideas to real-world devices.

Reaction rate is the fundamental quantity describing how fast a reactant is consumed or a product is formed. It is expressed as the change in concentration of a species per unit time, typically in units of $\text{mol} \cdot \text{L}^{-1} \cdot \text{s}^{-1}$. In a premixed methane-air flame, the rate of methane consumption can be written as $-d[\text{CH}_4]/dt$. The sign convention is important: A negative rate indicates reactant loss, while a positive rate indicates product formation.

Rate law (or kinetic expression) relates the reaction rate to the concentrations (or partial pressures) of the reacting species. For an elementary step, the rate law is directly proportional to the product of the reactant concentrations raised to the power of their stoichiometric coefficients. For example, the elementary reaction $\text{O} + \text{H}_2 \rightarrow \text{OH} + \text{H}$ has a rate law $r = k[\text{O}][\text{H}_2]$, where k is the rate constant. In many combustion mechanisms the overall rate law is a sum of contributions from many elementary steps, each with its own constant.

Reaction order indicates the exponent to which a concentration term appears in the rate law. For the previous elementary step the overall order is two (first order in O and first order in H_2). The order may be integer, fractional, or even zero. In a chain-branching reaction such as $\text{H} + \text{O}_2 + \text{M} \rightarrow \text{HO}_2 + \text{M}$, the presence of the third body M makes the reaction effectively third order, though the third-body efficiency can vary with the identity of M.

Overall order of a complex reaction is the sum of the orders of all reactants appearing in the rate law for the overall process. It is not necessarily equal to the number of reactants because intermediate species may appear in the expression after applying the steady-state approximation.

Zero-order kinetics occur when the rate is independent of the concentration of a particular reactant. This situation can arise in surface reactions where the active sites are saturated, as in catalytic combustion on a platinum surface at high fuel concentrations.

First-order reactions have a rate directly proportional to a single reactant concentration. Many unimolecular decompositions, such as the thermal dissociation of $\text{H}_2\text{O}_2 \rightarrow 2\text{OH}$, follow first-order behavior under low-pressure conditions.

Second-order reactions involve the product of two concentration terms. The classic example in combustion

is the recombination of two radicals, $\text{H} + \text{O} \rightarrow \text{OH}$, which proceeds with a rate proportional to $[\text{H}][\text{O}]$.

Partial order describes the exponent of a particular reactant in a multi-reactant rate law, which may differ from its stoichiometric coefficient. For instance, the reaction $\text{CO} + \text{OH} \rightleftharpoons \text{CO}_2 + \text{H}$ has a measured rate law $r = k[\text{CO}][\text{OH}]^{0.5}$, indicating a half-order dependence on OH.

Rate constant (k) is the proportionality factor in the rate law that encapsulates the intrinsic speed of an elementary step at a given temperature and pressure. The magnitude of k varies dramatically across reactions; in combustion, radical-radical recombination steps often have large k values (10^{13} – $10^{14} \text{ cm}^3 \cdot \text{mol}^{-1} \cdot \text{s}^{-1}$), whereas unimolecular decompositions may have smaller constants.

Arrhenius equation provides the temperature dependence of the rate constant: $k = A \exp(-E_a/RT)$. Here A is the pre-exponential factor, E_a is the activation energy, R is the universal gas constant, and T is the absolute temperature. The exponential term reflects the fraction of molecules possessing enough kinetic energy to overcome the energy barrier. In combustion, the high temperatures (often $> 1500 \text{ K}$) dramatically increase k values, accelerating radical production and chain branching.

Pre-exponential factor (A) incorporates the frequency of successful collisions and the orientation factor (steric factor). For a bimolecular reaction, A has units of $\text{cm}^3 \cdot \text{mol}^{-1} \cdot \text{s}^{-1}$; for a unimolecular reaction, it has units of s^{-1} . The magnitude of A can be estimated from transition-state theory, but in practice it is often obtained by fitting experimental data.

Activation energy (E_a) is the minimum energy that reacting molecules must acquire to reach the transition state. High activation energies correspond to slow reactions at low temperatures. In flame chemistry, the reaction $\text{H} + \text{O}_2 \rightarrow \text{OH} + \text{O}$ has an activation energy of about 40 kJ mol^{-1} , making it relatively fast at flame temperatures, whereas the isomerization of $\text{CH}_3\text{O}_2 \rightarrow \text{CH}_3\text{O} + \text{O}_2$ has a higher barrier and becomes significant only at elevated temperatures.

Collision theory explains reaction rates in terms of molecular collisions. For a reaction to proceed, colliding molecules must possess kinetic energy greater than E_a and the correct orientation. The theory predicts that the rate constant is proportional to the product of the collision frequency (Z) and the steric factor (ρ). Although collision theory provides a useful conceptual framework, it often overestimates rates because not all energetic collisions lead to reaction.

Transition-state theory (TST) refines collision theory by explicitly considering the formation of an activated complex (the transition state) that is in quasi-equilibrium with reactants. The TST expression for the rate constant is $k = (k_B T/h) \exp(-\Delta G^\ddagger/RT)$, where k_B is Boltzmann's constant, h is Planck's constant, and ΔG^\ddagger is the Gibbs free energy of activation. TST is widely used to calculate rate constants from quantum-chemical potential energy surfaces, especially for reactions where experimental data are scarce.

Reaction mechanism is a detailed description of the sequence of elementary steps that collectively produce the observed overall reaction. In combustion, mechanisms can involve hundreds of species and thousands of reactions. For example, a typical detailed mechanism for n-butane oxidation may contain over 500 species and 2 500 reactions, capturing low-temperature chain-branching pathways, high-temperature oxidation, and soot precursors.

Elementary step is a single molecular event that cannot be decomposed into simpler reactions. Each step has its own rate law that follows the law of mass action. In a mechanism, elementary steps may be reversible or irreversible. The step $\text{H} + \text{O}_2 + \text{M} \rightarrow \text{HO}_2 + \text{M}$ is an elementary three-body association, while its reverse, $\text{HO}_2 + \text{M} \rightarrow \text{H} + \text{O}_2 + \text{M}$, is a dissociation.

Chain-branching reaction creates additional radical species, increasing the overall reaction rate. The classic chain-branching step $\text{H} + \text{O}_2 \rightarrow \text{OH} + \text{O}$ is central to flame propagation because each H atom consumed yields two radicals (OH and O) that can further react. Chain branching is responsible for the rapid acceleration of reaction zones and determines the minimum ignition temperature.

Chain-propagation reaction sustains the radical population without changing the total number of radicals. Examples include $\text{O} + \text{H}_2 \rightarrow \text{OH} + \text{H}$ and $\text{OH} + \text{H}_2 \rightarrow \text{H}_2\text{O} + \text{H}$. These steps transfer radicals from one species to another, facilitating the continuous progress of combustion.

Chain-termination reaction removes radicals from the system, reducing the overall reactivity. Typical termination steps are radical recombination events such as $\text{H} + \text{H} \rightarrow \text{H}_2$ or $\text{OH} + \text{OH} \rightarrow \text{H}_2\text{O}_2$. In low-temperature combustion, termination reactions can dominate, leading to flame extinction.

Radical is an atom, molecule, or ion possessing an unpaired electron, making it highly reactive. Common combustion radicals include H, O, OH, CH₃, and HO₂. Radicals drive the rapid chemistry of flames, and their concentrations are often the most sensitive variables in kinetic simulations.

Steady-state approximation (SSA) assumes that the concentration of short-lived intermediates (usually radicals) remains essentially constant over the time scale of interest. Mathematically, $d[\text{intermediate}]/dt \approx 0$. This approximation simplifies the derivation of overall rate expressions for complex mechanisms. For instance, applying SSA to the HO₂ intermediate in the low-temperature oxidation of alkanes yields a reduced rate law that captures the net production of CO and H₂O.

Quasi-steady-state approximation (QSSA) is a more general form of SSA, applied when the intermediate's rate of change is small compared to the rates of formation and consumption. In combustion modelling, QSSA is often used for species like CH₃O₂ or H₂O₂, whose lifetimes are short relative to the flame residence time.

Rate-determining step (RDS) is the slowest elementary step in a reaction sequence, controlling the overall reaction rate. Identifying the RDS allows simplification of the mechanism and informs catalyst design. In many high-temperature hydrocarbon flames, the recombination of OH radicals ($\text{OH} + \text{OH} \rightarrow \text{H}_2\text{O}_2$) can become rate-determining, limiting the production of water.

Catalyst is a substance that accelerates a chemical reaction without being consumed. In combustion, catalytic surfaces such as Pt, Pd, or Rh are used in afterburners, catalytic combustors, and emission control devices. Catalysts lower the activation energy, modify the reaction pathway, and can enable oxidation at lower temperatures, reducing NO_x formation.

Inhibitor (or retardant) is a species that slows down a reaction, often by scavenging radicals. Halogenated compounds (e.g., CF₃Br) act as flame inhibitors because they readily capture H atoms, forming less reactive

radicals and thus suppressing chain branching.

Autocatalysis occurs when a product of a reaction serves as a catalyst for the same reaction, leading to an accelerating rate. In combustion, the formation of OH radicals can be autocatalytic because OH promotes further fuel oxidation, creating more OH in a positive feedback loop.

Laminar flame speed (S_L) is the speed at which a planar, one-dimensional premixed flame propagates into the unburned mixture under laminar flow conditions. It depends on the mixture composition, pressure, temperature, and kinetic parameters. Accurate prediction of S_L is crucial for designing stable burners and for assessing flame stability limits.

Turbulent flame speed (S_T) is the effective propagation speed of a flame in a turbulent flow field. It is typically larger than S_L due to flame wrinkling and increased surface area. Empirical correlations relate S_T to S_L , the turbulence intensity, and the integral length scale. Understanding S_T is essential for scaling laboratory data to industrial combustors.

Ignition delay (τ_{ig}) is the time interval between the start of heating (or pressure rise) and the onset of rapid exothermic reaction. In shock tube experiments, τ_{ig} is measured as the time between the reflected shock and the appearance of a marker species such as OH. Ignition delay is a sensitive probe of kinetic models, particularly for high-pressure, low-temperature fuel oxidation.

Ignition temperature (T_{ig}) is the minimum temperature at which a fuel-air mixture will spontaneously ignite under specified conditions. It is a function of pressure, mixture composition, and the presence of catalysts or inhibitors. For methane-air mixtures at atmospheric pressure, T_{ig} is around 540 °C, whereas for hydrogen-air it is roughly 560 °C, reflecting the high reactivity of H₂.

Flame temperature (T_f) is the temperature reached in the reaction zone after the fuel has been largely consumed. In adiabatic, stoichiometric hydrocarbon flames, T_f can exceed 2100 K. Real flames are cooler due to heat losses, incomplete combustion, and dilution. Flame temperature influences emission formation, material limits, and NO_x production.

Flame thickness (δ_f) characterizes the spatial extent of the high-temperature reaction zone. It is roughly proportional to the thermal diffusivity divided by the laminar flame speed ($\delta_f \approx \alpha/S_L$). Thin flames ($\delta_f \approx 0.5$ mm) are typical of lean hydrogen flames, while richer hydrocarbon flames can have thicker reaction zones.

Damköhler number (Da) is the ratio of the characteristic flow time to the chemical time scale, $Da = \tau_{flow}/\tau_{chem}$. In a well-mixed reactor, $Da \gg 1$ indicates that chemistry is fast compared to mixing, leading to near-equilibrium composition. In a combustor, $Da \approx 1$ marks the transition between flame-holding and flame-blowing-out regimes.

Peclet number (Pe) expresses the relative importance of convective transport to diffusive transport: $Pe = uL/\alpha$, where u is the flow velocity, L a characteristic length, and α the thermal diffusivity. High Peclet numbers imply that advection dominates, which is typical in high-speed gas turbines.

Reynolds number (Re) quantifies the ratio of inertial forces to viscous forces: $Re = \rho u L / \mu$, with ρ the density, μ the dynamic viscosity, and L a characteristic length. Turbulent combustion devices operate at $Re \gg 10^4$, where turbulent mixing strongly influences flame structure.

Lewis number (Le) is the ratio of thermal diffusivity to mass diffusivity: $Le = \alpha / D$. For many gases Le is close to unity, but for hydrogen Le is much greater. Flame stretch describes the combined effect of strain rate and curvature on flame propagation. The stretch rate (κ) is defined as the rate of change of flame area per unit area. Positive stretch generally reduces flame speed, while negative stretch can accelerate it. The flame's response to stretch is characterized by the Markstein length, which can be positive (stabilizing) or negative (destabilizing).

Markstein length (L_M) quantifies the sensitivity of flame speed to stretch. It is defined by the linear relation $S_L(\kappa) = S_L^0 - L_M \kappa$, where S_L^0 is the unstretched laminar flame speed. Positive L_M indicates that stretch reduces the flame speed, contributing to flame stability. For lean methane-air flames, L_M is typically positive, whereas hydrogen-air flames can exhibit negative L_M , making them more prone to cellular instabilities.

Thermal runaway is a situation where the exothermic reaction rate accelerates faster than heat can be removed, leading to a rapid temperature rise. In combustion, thermal runaway is the underlying mechanism of explosion. It is prevented by controlling heat loss, using appropriate dilution, and managing pressure.

Explosion limit defines the range of fuel-air mixtures that can sustain a self-propagating flame. The lower explosion limit (LEL) and upper explosion limit (UEL) are expressed in volume percent of fuel. For methane, the LEL is about 5% and the UEL about 15% at atmospheric pressure. Kinetic models help predict how these limits shift with temperature, pressure, and the presence of diluents.

One-step model simplifies combustion chemistry to a single global reaction, such as $CH_4 + 2 O_2 \rightarrow CO_2 + 2 H_2O$. The rate is expressed with an empirical rate constant that captures the overall reactivity. One-step models are useful for rapid CFD simulations where detailed chemistry would be computationally prohibitive, but they cannot predict intermediate species or pollutant formation.

Two-step model introduces an intermediate, often a "fuel radical" or "heat release" step, to improve fidelity. For example, a two-step model for hydrogen combustion may include $H_2 + O_2 \rightarrow 2 OH$ (slow) followed by $2 OH \rightarrow H_2O + O$ (fast). These models can capture ignition delay more accurately while retaining computational efficiency.

Detailed mechanism contains a comprehensive set of elementary steps and species, often derived from high-level quantum chemistry and validated against experimental data. Detailed mechanisms enable prediction of minor species, pollutant formation (NO_x , CO, soot), and flame extinction phenomena. However, they are computationally intensive, requiring reduction techniques for practical CFD integration.

Reduced mechanism is obtained by eliminating less important species and reactions from a detailed mechanism while preserving key kinetic features. Methods such as sensitivity analysis, principal component analysis, and directed relation graph (DRG) are used to identify and retain the most influential pathways. A reduced mechanism for gasoline surrogate fuels may contain 20-30 species and 100-200 reactions, offering

a good compromise between accuracy and cost.

Sensitivity analysis quantifies how changes in rate constants affect a target observable (e.g., Flame speed, ignition delay). The normalized sensitivity coefficient $S_{ij} = (\partial \ln Y) / (\partial \ln k_i)$ reveals which reactions dominate the behavior. In methane combustion, the sensitivity of S_L to the $H + O_2 \rightarrow OH + O$ step is typically large, indicating that accurate knowledge of this rate constant is critical.

Uncertainty quantification assesses the impact of uncertainties in kinetic parameters on model predictions. Techniques such as Monte Carlo sampling, Bayesian inference, and polynomial chaos expansion are employed. For high-pressure combustion, uncertainties in pressure-dependent rate constants can lead to large variations in predicted ignition delay, highlighting the need for robust experimental data.

Pressure-dependent reaction exhibits a rate constant that varies with pressure due to the involvement of a third body (M) or the formation of a chemically activated intermediate. The Lindemann, Troe, and SRI formulations provide expressions to model this dependence. An example is the recombination of $CH_3 + CH_3 + M \rightarrow C_2H_6 + M$, which proceeds faster at higher pressures because collisions stabilize the energized intermediate.

Third-body efficiency (α_M) reflects how effectively a particular species serves as a collision partner in a three-body reaction. In practice, efficiencies are expressed relative to a reference species (often N_2). For instance, in the reaction $O + H_2 + M \rightarrow OH + H + M$, O_2 may have an efficiency of 1.2, while Ar may have 0.7, indicating that O_2 is more effective at stabilizing the energized complex.

Fall-off region describes the transition between low-pressure (second-order) and high-pressure (first-order) limits of a pressure-dependent reaction. The reduced pressure parameter (Pr) quantifies the position within the fall-off region. Accurate modeling of fall-off is essential for predicting combustion behavior in engines that operate across a wide pressure range.

Modified strong collision (MSC) model and Troe formulation are advanced methods for describing the pressure dependence of unimolecular reactions. They introduce broadening parameters (F_{cent}) and temperature-dependent terms to capture the complex behavior observed experimentally. These models are incorporated into kinetic databases such as CHEMKIN and Cantera.

Equilibrium constant (K_{eq}) relates the forward and reverse rate constants via the relationship $K_{eq} = k_f/k_r = \exp(-\Delta G^0/RT)$. In combustion, equilibrium constants for reactions like $CO + H_2O \rightleftharpoons CO_2 + H_2$ are used to evaluate the reversibility of steps, especially at high temperatures where many reactions become quasi-reversible.

Thermodynamic consistency ensures that the set of forward and reverse rate constants obey the equilibrium constants derived from thermodynamic data. Inconsistent kinetics can lead to non-physical predictions such as negative concentrations or violation of the second law. Software tools enforce consistency by adjusting rate constants to match the thermodynamic database.

Adiabatic flame temperature is the temperature attained by a flame assuming no heat loss to the surroundings. It is calculated by equating the enthalpy of reactants to that of products, using the

appropriate specific heats. The adiabatic temperature provides an upper bound for the actual flame temperature and is used in performance analysis of gas turbines.

Heat release rate (HRR) quantifies the rate at which chemical energy is converted to thermal energy per unit volume. In CFD simulations, HRR is a source term in the energy equation. Accurate prediction of HRR requires reliable kinetic models, because the rate of heat release is directly linked to the reaction rates of the dominant pathways.

Flamelet is a concept used in turbulent combustion modelling, where the flame is treated as an ensemble of locally one-dimensional laminar flame structures (flamelets) embedded in a turbulent flow. Flamelet libraries store pre-computed solutions for species and temperature as functions of mixture fraction and scalar dissipation rate, enabling rapid evaluation during CFD runs.

Scalar dissipation rate (χ) measures the rate at which mixture fraction gradients are smoothed by molecular diffusion. In flamelet theory, χ controls the thickness of the flamelet and thus the local extinction and reignition behavior. High χ values correspond to strong strain, which can quench the flame.

Extinction occurs when the flame cannot sustain the reaction due to excessive stretch, dilution, or heat loss. In practical combustors, extinction leads to flame blow-out, loss of power, and potentially hazardous conditions. Kinetic models predict extinction limits by evaluating when the net production of radicals falls below a critical threshold.

Reignition is the process by which a previously extinguished flame re-establishes after a favorable change in conditions (e.g., reduced strain or increased temperature). Reignition is governed by the availability of radicals and the balance between chain-branching and termination steps. In scramjet engines, rapid reignition after shock-induced extinction is a design challenge.

NO_x formation in combustion primarily occurs via the thermal (Zeldovich) mechanism, which involves the high-temperature reactions $\text{N}_2 + \text{O} \leftrightarrow \text{NO} + \text{N}$ and $\text{N} + \text{O}_2 \leftrightarrow \text{NO} + \text{O}$. The rate of NO formation is extremely temperature-sensitive, scaling roughly with $\exp(-E_a/RT)$ where $E_a \approx 300 \text{ kJ mol}^{-1}$. Accurate kinetic modeling of NO_x requires inclusion of these high-activation-energy steps and consideration of residence time.

CO formation arises from incomplete oxidation of carbon-containing species. The primary pathway is $\text{CO} + \text{OH} \leftrightarrow \text{CO}_2 + \text{H}$, which can be limited by insufficient OH radicals at low temperatures. In rich combustion, CO can accumulate because the oxidation chain is interrupted, leading to higher emissions.

Soot precursor chemistry involves the formation of polycyclic aromatic hydrocarbons (PAHs) from small hydrocarbon radicals. The HACA (hydrogen-abstraction- C_2H_2 -addition) mechanism describes the stepwise growth of PAHs, ultimately leading to soot particles. Soot formation is highly sensitive to fuel structure, temperature, and residence time, and kinetic models must incorporate detailed PAH pathways to predict particulate emissions.

Hydrogen-air combustion is often used as a benchmark system because of its simple chemistry and high reactivity. The key radical pool consists of H, O, and OH, and the dominant chain-branching step is $\text{H} + \text{O}_2$

→ OH + O. Because hydrogen has a low molecular weight, its diffusion coefficient is high, leading to a Lewis number less than one and pronounced preferential diffusion effects.

Hydrocarbon fuel oxidation proceeds through a complex network of reactions, including H-abstraction, addition to O₂, formation of peroxy radicals, isomerization, and β-scission. For example, the oxidation of ethane begins with C₂H₆ + H → C₂H₅ + H₂, followed by C₂H₅ + O₂ → C₂H₅O₂, which can then isomerize to CH₃CH₂O₂ → CH₃CHO + OH. Understanding each of these steps is essential for predicting ignition delay and flame speed for liquid fuels.

Low-temperature oxidation of hydrocarbons, relevant for gasoline and diesel fuels, involves the formation of hydroperoxyalkyl radicals (QOOH) and subsequent chain-branching pathways that can cause ignition delays on the order of milliseconds. The “negative temperature coefficient” (NTC) behavior, where the reaction rate decreases with increasing temperature over a certain range (≈ 600–800 K), is a hallmark of this regime and poses a challenge for kinetic model validation.

High-temperature oxidation dominates above ~1200 K, where the peroxy chemistry becomes less important and direct H-abstraction and radical recombination control the reaction rate. In this regime, the Arrhenius temperature dependence is more pronounced, and the flame speed is largely dictated by the balance between chain-branching and termination steps.

Pressure-dependent chemistry becomes increasingly important in gas-turbine combustors, where pressures can exceed 30 bar. At high pressure, three-body recombination reactions are enhanced, altering the radical pool and shifting the balance between branching and termination. For instance, the formation of HO₂ via H + O₂ + M → HO₂ + M is dramatically accelerated, influencing the low-temperature ignition behavior of fuels.

Transport properties such as viscosity, thermal conductivity, and mass diffusivity affect the coupling between chemistry and flow. In CFD, these properties are needed to solve the Navier-Stokes equations and to compute diffusion fluxes. Accurate transport data are especially critical for light species like H₂, where diffusion rates are high and influence flame structure.

Effective activation energy is an empirical parameter obtained by fitting the overall reaction rate to an Arrhenius form, even when the underlying mechanism consists of many steps with different activation energies. The effective activation energy can vary with temperature and pressure, reflecting the changing contribution of individual pathways.

Reaction path analysis visualizes the flow of carbon through a mechanism, identifying the most important routes from fuel to products. By quantifying the fraction of fuel carbon that passes through each intermediate (e.g., C₂H₅, C₃H₇, etc.), Engineers can target specific reactions for reduction or improvement, focusing computational resources on the most influential steps.

Rate of production analysis (ROPA) evaluates the net creation or destruction of each species by each reaction. This technique reveals which reactions dominate the formation of key radicals or pollutants. In a methane flame, ROPA typically shows that the H + O₂ → OH + O step is the primary source of OH, while the OH + H₂ → H₂O + H step is the main sink of H.

Flame stability is a multifaceted concept encompassing the ability of a flame to resist extinction under varying flow, pressure, and mixture conditions. Stability criteria often involve the balance between heat release and heat loss, quantified by the flame–flow interaction parameter ($S = S_L/u$). A stable flame requires $S > 1$ in a laminar flow, whereas turbulent flames depend on the turbulent Reynolds number and Damköhler number.

Flame stretch factor (K) combines the effects of curvature (κ_c) and strain (κ_s) into a single parameter: $K = \kappa_c + \kappa_s$. Positive K reduces the local flame speed, while negative K can increase it. In practical burners, geometric features such as bluff bodies introduce curvature, and inlet velocity gradients produce strain, both of which must be accounted for in design.

Markstein number (Ma) is the dimensionless counterpart of the Markstein length, obtained by dividing L_M by the flame thickness. It provides a convenient way to compare the stretch sensitivity of different fuels. A higher Ma indicates greater sensitivity, which can lead to cellular flame structures in hydrogen-rich mixtures.

Flame anchoring refers to the mechanism by which a flame remains fixed at a location, often due to a heat source, recirculation zone, or solid surface. Anchoring is crucial in industrial burners to prevent flame blow-out. Kinetic considerations include the local temperature rise provided by the anchor, which must be sufficient to sustain chain-branching reactions.

Flame quenching distance is the minimum distance between a flame and a cold surface at which the flame can be extinguished. It depends on the flame thickness and the thermal diffusivity of the gas. For hydrogen flames the quenching distance is on the order of 0.2 Mm, whereas for methane it is about 0.5 Mm. Accurate prediction of quenching distances informs safety standards for equipment spacing.

Flame holder devices, such as cavities or bluff bodies, create recirculation zones that provide hot gases to stabilize the flame. The recirculation enhances the residence time of radicals, effectively increasing the local Damköhler number and promoting stable combustion. Design of flame holders must consider both aerodynamic losses and the impact on flame stretch.

Premixed versus non-premixed combustion distinguishes whether fuel and oxidizer are mixed before ignition (premixed) or meet at the flame front (non-premixed). Kinetic models for premixed flames focus on homogeneous reaction zones, whereas non-premixed flames require detailed modeling of diffusion-controlled reaction layers, often using flamelet or partially premixed approaches.

Partial premixing occurs in many practical combustors where fuel and air are mixed to a limited extent before ignition. This regime combines features of both premixed and diffusion flames, leading to complex flame structures. Kinetic modeling of partially premixed flames must capture the interaction between local equivalence ratio and scalar dissipation rate.

Flamelet generated manifolds (FGM) extend the flamelet concept by constructing manifolds that map a reduced set of variables (e.g., Mixture fraction and progress variable) to thermochemical states. FGMs enable efficient integration of detailed chemistry into CFD while preserving the influence of strain and extinction on the flame.

Progress variable (C) is a scalar that quantifies the advancement of combustion, often defined as a linear combination of species mass fractions (e.g., $C = Y_{\text{CO}_2} + Y_{\text{H}_2\text{O}}$). In reduced-order models, the source term for C is derived from the detailed mechanism, allowing the flame to be represented by a single transport equation.

Time-scale separation is the principle that chemical reactions can be categorized into fast, moderate, and slow processes. Fast reactions (e.g., Radical recombination) reach quasi-steady state quickly, while slow reactions (e.g., Fuel oxidation at low temperature) control the overall progress. Exploiting this separation enables model reduction through methods such as ILDM (intrinsic low-dimensional manifold).

Intrinsic low-dimensional manifold (ILDM) is a mathematical technique that reduces the dimensionality of the chemical kinetic system by identifying a low-dimensional surface on which the slow dynamics evolve. ILDMs are constructed by solving for the fast modes (eigenvectors with large negative eigenvalues) and eliminating them, yielding a compact representation suitable for CFD.

Computational stiffness arises when a system of ordinary differential equations contains widely disparate time scales, forcing the numerical solver to adopt very small time steps to maintain stability. Combustion kinetics is notorious for stiffness due to the coexistence of fast radical reactions and slow fuel oxidation steps. Implicit solvers (e.g., VODE, CVODE) and specialized integration algorithms are employed to manage stiffness.

Operator splitting separates the solution of transport (advection, diffusion) from chemistry (reaction) within each time step. The Strang splitting method, for example, alternates half-steps of chemistry with a full step of transport, achieving second-order accuracy. Operator splitting enables the use of specialized solvers for each sub-problem, improving computational efficiency.

Adaptive chemistry dynamically selects the appropriate level of kinetic detail based on local conditions, activating a detailed mechanism in regions of high reactivity (e.g., Flame front) while using a reduced mechanism elsewhere (e.g., Downstream of the flame). This approach reduces computational cost without sacrificing accuracy in critical zones.

Stiffness reduction techniques include the use of quasi-steady-state approximations for selected radicals, lumping of similar species, and tabulation of reaction rates. These methods aim to decrease the disparity between the fastest and slowest time scales, allowing larger integration steps.

Monte Carlo uncertainty analysis propagates uncertainties in kinetic parameters through a combustion model by random sampling, generating a distribution of outcomes (e.g., Flame speeds). The resulting confidence intervals inform the reliability of predictions and identify which parameters most influence model uncertainty.

Bayesian inference updates prior probability distributions of kinetic parameters using experimental data, yielding posterior distributions that reflect both prior knowledge and new evidence. Bayesian methods are increasingly applied to calibrate reaction rate constants for complex fuel mechanisms, leveraging data from shock tubes, rapid compression machines, and laminar flame speed measurements.

Thermal NO formation follows the extended Zeldovich mechanism, which includes the reactions $N_2 + O \leftrightarrow NO + N$, $N + O_2 \leftrightarrow NO + O$, and $N + OH \leftrightarrow NO + H$. The rate of NO formation is highly temperature dependent, and the production peaks near 2100 K. Suppressing thermal NO often involves staging the combustion to keep peak temperatures below the NO formation threshold.

Prompt NO formation occurs through the Fenimore mechanism, wherein hydrocarbon radicals react with atmospheric nitrogen: $R + N_2 \rightarrow RN + N$. The resulting nitrogen atoms can then form NO. Prompt NO becomes significant in fuel-rich flames where hydrocarbon radicals are abundant.

Soot modeling incorporates both nucleation and surface growth processes. The nucleation step forms primary particles from PAH clusters, while surface growth adds carbon layers via reactions such as $C_2H_2 + \text{radical} \rightarrow \text{soot}$. Models may also include oxidation ($\text{soot} + O_2 \rightarrow CO_2$) and agglomeration (coagulation of particles). Accurate soot prediction requires coupling of kinetics with turbulence-induced particle transport.

Particle size distribution is often described by a log-normal function, characterized by a geometric mean diameter and a standard deviation. In combustion simulations, the particle size distribution influences radiative heat transfer, deposition rates, and filter performance.

Radiative heat transfer in combustion systems is governed by the spectral properties of gases (e.g., H_2O , CO_2) and soot particles. The rate of radiative loss affects flame temperature and, consequently, reaction rates. Detailed kinetic models must be paired with radiative models to capture the feedback between chemistry and thermal radiation.

Heat loss through walls, exhaust, or radiation reduces the flame temperature, thereby slowing the reaction rates. In practical burners, heat loss is mitigated by insulating materials, recirculation of hot gases, and optimized geometry. The impact of heat loss on kinetic predictions is quantified by the heat loss parameter ($\beta = Q_{\text{loss}}/Q_{\text{release}}$).

Flame stretch and extinction limit are closely linked: A flame subjected to excessive stretch (high K) may reach a critical stretch rate (K_{crit})